# Probing of micro and nanocapsules properties with instrumented indentation A.E. Giannakopoulos<sup>1\*</sup>, V.I. Zafiropoulou<sup>1</sup>, Th. Zisis<sup>2</sup>, A.D. Lytra<sup>3</sup> and N.A. Pelekasis<sup>3</sup>

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### Introduction

Contrast agents (CA) are microbubbles covered with an elastic shell, usually made of polymeric or lipid biomaterials providing mechanical strength and decelerating the gas dissolution in vivo. Their initial diameter ranges from 3 to 5 µm and their thickness from 10 to 40 nm<sup>1-2</sup>. CA microbubbles have been successfully used the last decades for imaging of human organs and they are considered as drug/gene carriers for therapeutic application of diseases. In both cases the elastic properties of the shell are the key parameters that control their behavior in ultrasound environment through arteries and tissues. The present work aims at:



- The estimation of Young's and bending modulus
- The numerical investigation of their static response (FEM & Abaqus)

## **Theoretical Formulation**

ext

 $\mathsf{P}_{\mathsf{G}}$ 

ξ=1



Normal and Tangential Force Balance<sup>3</sup>

$$\Delta F_{n} = k_{s}\tau_{s} + k_{\varphi}\tau_{\varphi} - \frac{1}{\sigma}\frac{\partial(\sigma q)}{\partial s} = \Delta P, \qquad \Delta F_{s} = -\left(\frac{\partial\tau_{s}}{\partial s} + \frac{1}{\sigma}\frac{\partial\sigma}{\partial s}(\tau_{s} - \tau_{\varphi}) + k_{s}q\right) = 0$$
$$q = \frac{1}{\sigma}\frac{\partial\sigma}{\partial s}\left(\frac{\partial(\sigma m_{s})}{\partial s} - m_{\varphi}\right), \qquad \sigma = r\sin\theta$$



**Isothermal Gas Compression** 

 $P_G V_f = P_A V_i$ 

**Position of Mass Center** 

$$\mathbf{z}_{cm} = \frac{1}{V} \bigoplus_{v} \vec{\mathbf{r}} \cdot \vec{\mathbf{e}}_{z} dV = 0$$

#### **Kinematic Condition**

$$r(\xi = 0) = r\cos\theta$$
 for  $\xi \in [0, \xi_c]$ 

#### **Constitutive Law<sup>6</sup>**

$$\tau_{i} = G_{s} \frac{1 + \nu}{1 - \nu} (\lambda_{i}^{2} - 1), \qquad m_{i} = \frac{k_{b} (K_{i} + \nu K_{j})}{\lambda_{j}}$$
$$K_{i} \equiv \lambda_{i} k_{i} - k_{i}^{R}, \qquad k_{b} = \frac{Eh^{3}}{12(1 - \nu^{2})}, \qquad \lambda_{i} = \frac{ds_{i}}{dS_{i}} \quad i, j = s \text{ or } \varphi$$

**Boundary Conditions** 

$$\begin{aligned} \mathbf{r}_{\xi} &= 0 & \text{and} & \theta_{\xi\xi} &= 0 & \text{at} & \xi &= 0 & \text{and} & \xi &= 1 \\ \mathbf{P}_{\mathsf{ext},\,\xi} &= 0 & \text{and} & \mathbf{P}_{\mathsf{ext},\,\xi\xi} &= \mathbf{L}_{\xi\xi} & \text{at} & \xi &= 0 \\ \mathbf{P}_{\mathsf{ext},\,\xi} &= \mathbf{L}_{\xi} & & \text{at} & \xi &= \xi_{\mathsf{c}} \end{aligned}$$

## **Numerical Implementation**

| k <sub>c</sub> =0.61 N/m | Experimental<br>Values |        | Asymptotic Estimation<br>Point Load <sup>7</sup> |        | Asymptotic Estimation<br>Plane Contact |        |
|--------------------------|------------------------|--------|--|--------|--|--------|
| D <sub>o</sub> [µm]      | E [MPa]                | h [nm] | E [Mpa]  | h [nm] | E [MPa]                                | h [nm] |
| 2.6                      | 10-16                  | 20     | 8.5  | 25     | 20                                     | 16     |
| 3.5                      | 4-8                    | 26     | 12   | 26     | 5                                      | 26     |
| 4.1                      | 2.5-6                  | 31     | 6.1  | 31     | 4.6                                    | 33     |
| k <sub>c</sub> =1.14 N/m |                        |        |  |        |  |        |
| 3.1                      | 6-10                   | 23     | 3.4  | 35     | 10                                     | 19     |
| 3.2                      | 6-10                   | 24     | 14   | 20     | 6.7                                    | 27     |
| 4.0                      | 2.5-6                  | 30     | 4.7  | 30     | 4.7                                    | 28     |
| 4.9                      | 1-3                    | 37     | 4.5  | 31     | 4.9                                    | 28     |
| 5.5                      | 1-3                    | 41     | 1.7  | 47     | 2                                      | 40     |

## Conclusions

- 1. It is possible to estimate both Young's modulus and shell thickness by a single forcedeformation (f-d) curve.
- 2. The point load can qualitatively predict the f-d curve, while the abaqus calculation also predicts transition from a linear (Reissner) to a non linear (Pogorelov) regime.
- 3. Reliable estimates of the shell thickness and elasticity modulus of polymeric shells are obtained in this fashion

#### Finite Elements with B-Cubic Splines as Basis Functions<sup>4</sup>

$$\mathbf{r}(\boldsymbol{\xi}) = \sum_{i=0}^{N+1} \mathbf{a}_{i} \mathbf{B}_{i}, \qquad \boldsymbol{\theta}(\boldsymbol{\xi}) = \sum_{i=0}^{N+1} \mathbf{b}_{i} \mathbf{B}_{i} \qquad \text{and} \qquad \mathbf{P}_{\mathsf{ext}}(\boldsymbol{\xi}) = \sum_{i=0}^{M+1} \mathbf{c}_{i} \mathbf{B}_{i}$$

#### **Newton-Raphson Iterations**

$$\mathbf{J}_{ij}(\mathbf{x}_{i}^{k})\Delta\mathbf{x}_{j}^{k} = \mathbf{R}_{i}^{k}(\mathbf{x}_{i}^{k}), \qquad \mathbf{J}_{ij} = \frac{\partial \mathbf{R}_{ij}}{\partial \mathbf{x}_{j}} \approx \frac{\mathbf{R}_{i}(\mathbf{x}_{j} + \Delta \mathbf{x}_{j})}{\Delta \mathbf{x}_{j}}, \qquad \Delta \mathbf{x}_{j}^{k} = \mathbf{x}_{j}^{k} - \mathbf{x}_{j}^{k+1}$$

# **Asymptotic Analysis**

#### Point Load<sup>5-6</sup>



#### **Plane Contact**

Stage I 
$$F = \frac{4}{\sqrt{3(1-v^2)}} \frac{Eh^2 \Delta}{R_o} + 0.06543 \frac{Eh\Delta^2}{R_o};$$
 Stage II  $F = 3.807 \frac{Eh^3}{(1-v^2)R_o} \left(\frac{\Delta \sqrt{1-v^2}}{h}\right)^{1/2}$ 

- 4. Relaxing the assumption of a point load provides reliable estimates for the pressure distribution, in accordance with previous investigations<sup>12</sup>, and predicts reduced displacements when crater formation takes place, in comparison with the point load distribution.
- 5. Consideration of the surface tension delays the transition from Reissner to Pogorelov regime.
- 6. Proper accounting of adhesion forces is expected to capture the response of phospholipid shells by eliminating the Pogorelov regime in favor of the gas compressibility dominated regime in the f/d curve at large external loads.

### **Selected References**

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